SUPPLEMENT TO EC130B4 RFM

AIR COMM CORPORATION 3300 AIRPORT ROAD BOULDER, COLORADO 80301

EUROCOPTER EC130B4

CABIN AIR CONDITIONING SYSTEM

FLIGHT MANUAL SUPPLEMENT

Document No. EC130B4-2

FAA APPROVED

The information contained in this document is FAA approved material which must be carried in the basic Flight Manual after the rotorcraft has been modified by installation of the Cabin Air Conditioning System in accordance with Air Comm Corporation STC No. SR00543DE.

This document is applicable to the EC130-202 Air Conditioning system (Condenser located in the LH or RH baggage compartment).

The information in this document supplements or supersedes the basic manual only in the items contained herein. For Limitations, Procedures, and Performance Data not contained in this supplement consult the basic Flight Manual.

FAA Approved Da Grossm

Ronald F. May, Manager
Denver Aircraft Certification Office
Northwest Mountain Region
Denver, Colorado

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CABIN AIR CONDITIONING SYSTEM

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			Ronald F. May, Manager
			Denver Aircraft Cert. Office
			Northwest Mountain Region
			Denver, Colorado
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CABIN AIR CONDITIONING SYSTEM

SECTION 1

INTRODUCTION

The EC130 air conditioner is a vapor cycle system that includes the following components:

Compressor Condenser Forward Mounted Evaporator Aft Mounted Evaporator (Optional Fresh Air Inlet) Plumbing System Electrical System

The compressor is belt driven through an electric clutch by a sheave mounted on the output shaft of the main rotor transmission.

The condenser coil and blower assembly are mounted in the left hand (optional RH) baggage compartment inside a metal enclosure. The outlet air is exhausted thru a cutout in the baggage door.

The forward evaporator is mounted on the forward end of the instrument panel console structure. Conditioned air is delivered to the crew by means of two air ducts, one mounted on each side of the instrument panel console.

The aft evaporator assembly is mounted on the right hand side of the main rotor transmission deck aft of the cabin bulkhead and is enclosed by the main rotor transmission cowling. Cabin return air is ducted to the evaporator through a cutout in the aft cabin bulkhead. Conditioned air is pumped to the headliner ducting through four inlets in the aft cabin bulkhead.

An optional fresh air assembly is available. This system features an actuator operated door in the aft evaporator, which can be positioned to either recirculate existing cabin air or to bring in outside fresh air. This door is typically positioned to recirculate existing cabin air during air conditioner operations to maximize cooling performance, but can be opened during air conditioner or fan only operations if desired.

The air conditioner system controls feature A/C - OFF - FAN functions incorporated into a single "three position" switch. Two fan speed switches are provided to control the forward and aft evaporator fans, which can be operated independently of each other. A temperature control knob is provided to vary the output air temperature of the air conditioner. A dimmer control knob is provided to vary the back lighting intensity of the air conditioner control panel.

A blue compressor "ON" light is located on the instrument panel to provide a visual status of the compressor operation.

The refrigerant plumbing system features high and low pressure cutoff switches. Exceeding the pressure limits will de-energize the compressor's magnetic clutch.

The air conditioner system features an auto load shed circuit that will de-energize the entire air conditioner system, including the compressor clutch, in the event of a generator/engine failure.

A "GND MAINT" switch, located on the AC Relay Panel in the aft equipment bay, is provided to allow maintenance personnel the means of powering the air conditioning electrical system when the engine/generator is off-line. Pressing the "GND MAINT" switch latches a relay that overrides the air conditioner auto-load-shed circuit. The relay will unlatch when the generator is turned on following an engine start.

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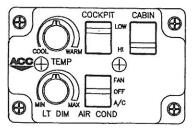
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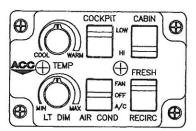
SECTION 2

OPERATING LIMITATIONS

PLACARDS AND MARKINGS



AC Control Panel without Fresh Air Switch



AC Control Panel with Fresh Air Switch

Figure 2
AC Control Panel located in cockpit headliner
(Alternate Location – Anywhere on RH side of instrument panel)

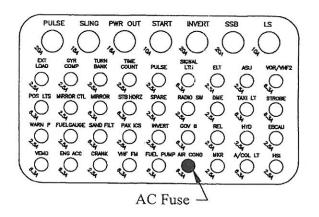


Figure 3
AC fuse is located in the cockpit fuse panel on the RH side of the instrument panel console

Aft Evop Inet (Located on RH Side)

Figure 1
General Arrangement – Cabin Air Conditioner

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PLACARDS AND MARKINGS (cont'd)

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PLACARDS AND MARKINGS (cont'd)

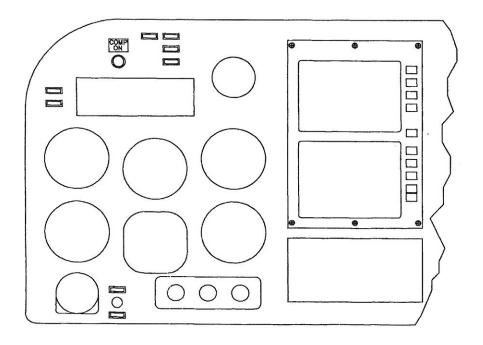
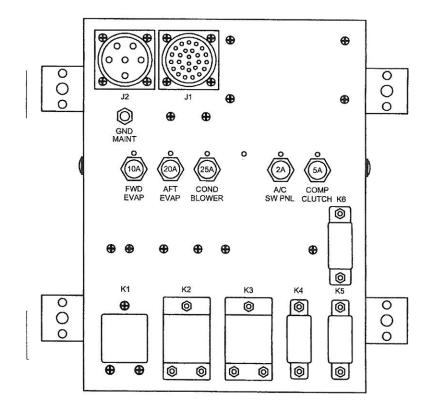


Figure 4
COMP ON Annunciator
(Approved Location – Anywhere on the instrument panel within view and reach of the pilot)

Figure 5
AC Relay Panel
(Mounted inside the right-hand baggage compartment)



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SECTION 3

EMERGENCY PROCEDURES

SECTION 4

NORMAL PROCEDURES

Place the A/C-OFF-FAN (3 position switch) to the OFF position for any of the following emergencies:

Smoke in the cabin Engine failure Engine over-temperature Generator failure Water landing

NOTE

Loss of generator output will activate the air conditioner auto load shed circuitry, which will de-energize the entire air conditioning system, including compressor clutch.

NOTE

If outlet air is not cool, place the A/C-OFF-FAN (3-postion switch) to the OFF or FAN position to preclude damage to the compressor.

PREFLIGHT CHECK (EXTERIOR)

Compressor – Check security Compressor Drive Belt – Check tension and general condition Compressor Belt Shield – Check security

ENGINE PRESTART Check A/C-OFF-FAN (3-postion switch) - OFF

BEFORE TAKEOFF

A/C-OFF-FAN (3-postion switch) – As desired EVAP FANS – FAN SPEED SWITCH – As desired FRESH AIR SWITCH – As desired

IN FLIGHT OPERATIONS

A/C-OFF-FAN (3-postion switch) – As desired EVAP FANS – FAN SPEED SWITCH – As desired FRESH AIR SWITCH – As desired

NOTE

Total air conditioning system electrical load is less than 43 amps. Monitor amps.

NOTE

Placing the FRESH/RECIRC switch in the FRESH position allows outside air to enter the aft evaporator. When maximum cooling is desired place this switch in the RECIRC position.

NOTE

Simultaneous operation of the cabin heater and air conditioner can be used to achieve cabin defogging.

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SECTION 5

PERFORMANCE DATA

When the air conditioner is operating, the performance data in the basic flight manual should be reduced as shown below:

Rate of Climb Degradation

Reduce the rate of climb in the basic Flight Manual by the amount shown below:

R/C Reduction 46 ft/min (14 m/min)

Hover Ceiling In Ground Effect and Out of Ground Effect

Add 40 lb (18 kg) to the actual IGE/OGE hover gross weight for takeoff power or maximum continuous power when entering the chart to determine hover ceiling.