

**BELL HELICOPTER
MODEL 206B
250-C20B/C20J ENGINE**

**FAA APPROVED
FLIGHT MANUAL SUPPLEMENT
FOR
CABIN COOLING, CABIN HEATING, OR
ENVIRONMENTAL CONTROL SYSTEM**

206-0100 or 206-0102

The information contained in this document is FAA approved material, which must be carried in the basic Flight Manual, dated July 1, 1977, after the rotorcraft has been modified by installation of Cabin Cooling, Cabin Heating, or the ECS system in accordance with Keith Products, Inc. STC SH1504NM.

The information in this document supplements or supersedes the basic manual only in the items contained herein. For Limitations, Procedures, and Performance Data not contained in this supplement, consult the basic Flight Manual.

Revised: August 3, 1995

LOG OF PAGES 206B FLIGHT MANUAL C20B/C20J ENGINE FAA APPROVED SUPPLEMENT

CABIN COOLING, CABIN HEATING OR ECS

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Original. . . . 0 . June 23, 1982

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FAA APPROVED DATE: June 23, 1982

Approved: J. P. Chudy for
Mark E. Baldwin, Manager
Denver-Aircraft Certification Field Office
Northwest Mountain Region, Aurora, Colorado

Revised: August 3, 1995

LOG OF Revisions 206B FLIGHT MANUAL
C20B/C20J
ENGINE

FAA APPROVED
SUPPLEMENT

CABIN COOLING, CABIN HEATING OR ECS

LOG OF REVISIONS			
NO.	REVISION DATE	PAGES REVISED	FAA APPROVAL
0	June 23, 1982	Orig. Issue	J.P. Chudy
1	November 1, 1982	All	J.P. Chudy
2	December 21, 1982	All	George H. Meyers III
3	March 20, 1984	All	George H. Meyers III
4	March 21, 1985	2,3,4,5,10	George H. Meyers III
5	May 5, 1989	1,2,3,6	Richard S. Adler
6	October 24, 1990	1,2,3	Larry M. Kelly
7	March 9, 1992	1,2,3,5,6	Gary B. Roach
8	April 10, 1995	1,2,3	James R. Arnold
9	April 18, 1995	2,3,6	James R. Arnold
10	August 3, 1995	All	<i>[Signature]</i>

NOTE: Revisions are indicated by a black vertical line.

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250-C20B/C20J ENGINE

CABIN COOLING, CABIN HEATING or ECS

FAA Approved

Date: June 23, 1982

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The Environmental Control System consists of a vapor cycle air conditioner for cabin cooling and cabin heating provided by engine bleed air. Cabin cooling and/or heating only may also be installed. Operating instructions contained herein are given separately for cabin cooling and/or heating and apply to the system installed.

The air conditioner system consists of an engine driven compressor, a condenser and evaporator(s), each of the latter using electric motor driven blowers. AIR COND or FAN may be selected. In the AIR COND mode, the evaporator blower(s) is(are) automatically activated, distributing cold air. The FAN mode permits cabin air circulation only, either in a HI or LO blower speed as selected by the second switch. Temperature control is accomplished through a rheostat to set desired cooling air temperature.

The heater control consists of a single (OFF/MAX) rotary control. The desired temperature is selected by a proportional adjustment of cabin heat (OFF/MAX) knob. When heater is on, the environmental control switch must be turned to FAN.

Defogging is best accomplished by simultaneous operation of the air conditioning and/or fresh outside air as heat is applied.

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For 206B models designated TH67 with Two Pilot IFR Configuration: The air conditioning equipment is powered by the non-essential bus and is protected by a 50 amp current limiter. A relay has been installed to automatically disconnect the air conditioning system in the event of a main generator failure.

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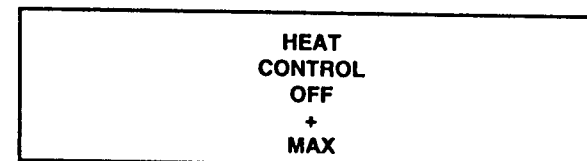
FAA Approved
Date: June 23, 1982
Revised: August 3, 1995**SECTION I: OPERATING LIMITATIONS****ELECTRICAL LOAD LIMITATIONS**

Maximum - 70% on Load meter as stated in basic Flight Manual.

For 206B models designated TH67 with the two pilot IFR Configuration only: the air conditioning system should not be selected to "Air Cond" when load meter exceeds 46%.

WEIGHT/CG LIMITATIONS

Weight change shall be calculated after kit is installed and ballast readjusted, if necessary, to return empty weight CG within allowable limits.

PLACARDS AND MARKINGS:

(Located on overhead next to circuit breaker panel when ECS or Heating only is installed)

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SECTION I: (Cont'd) OPERATING LIMITATIONS

PLACARDS AND MARKINGS (Cont'd)

**MAGNETIC COMPASS
IS UNRELIABLE
WITH AIR COND
OR FAN ON**

(When forward blower(s) installed)

**MAX BAGGAGE 170 LBS. (ECS ONLY)
MAX BAGGAGE 241 LBS. (CABIN HEATING ONLY)
MAX BAGGAGE 176.5 LBS. (CABIN COOLING ONLY)**

(Located on the inner surface of baggage compartment door when installed.)

**AIR CONDITIONER
+
A/C +
FAN**

(For 206B models: Located in circuit breaker panel area when ECS or Cabin Cooling only is installed.) (For 206B models designated TH67 with 2 pilot IFR configuration: Located in center cockpit pedestal on two seat IFR Aircraft when cabin cooling is installed)

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SECTION I: (Cont'd) OPERATING LIMITATIONS

PLACARDS AND MARKINGS (Cont'd)

**NOTE: FOR AIRCRAFT EQUIPPED WITH GROUND SEEKING
ANNUNCIATOR PANEL THE FOLLOWING CIRCUIT BKR NOT
REQUIRED.**

**HTR.
+
DUCT LT**

(Located in the circuit bkr. panel area or at the aft cabin buss when ECS or Cabin Heating only is installed.)

DUCT TEMP HIGH

(Located on instrument panel in caution light panel when ECS or Cabin Heating only is installed.)

**DO NOT SWITCH AIR COND.
ON WHEN LOAD METER
EXCEEDS 46%**

(Located on center cockpit pedestal on for 206B model designated TH67 with two pilot IFR configuration when cabin cooling is installed.)

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Revised: August 3, 1995**SECTION 2: NORMAL PROCEDURES****PREFLIGHT CHECK (EXTERIOR)**

5. Fuselage - Aft Left Side
Compressor - check security
Compressor drive belt - check general condition
and security

(When ECS or Cabin Cooling only is installed.)

ENGINE PRESTART CHECKEnvironmental Control Switch - OFF
Cabin Heat Valve - OFF**NOTE:**

The heat control pilot valve is a continuously variable one turn valve. The OFF position is fully counterclockwise.

**ENGINE RUN-UP CHECK, BEFORE TAKE-OFF, or
IN-FLIGHT OPERATIONS****NOTE:**

TOT increases with the bleed air heater and/or cabin cooling operating. Observe turbine outlet temperature limitations.

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Date: June 23, 1982
Revised: August 3, 1995**SECTION 2: (Cont'd) NORMAL PROCEDURES****CABIN COOLING**Environmental Control Switch - AIR COND
Cabin heat Valve - OFF (if installed)
Fan Switch - HI or LO (as desired)
Outlets - Close overhead windshield outlets and bottom
outlets on instru. panel side ducts.
- Open overhead outlets and aft outlet on instru.
panel ducts as desired.**CABIN HEAT**Environmental Control Switch - FAN (if installed)
Cabin Heat Valve - ON (adjust to desired level)
Fan Switch - HI or LO (as desired, if installed)**CABIN DEFOGGING**Environmental Control Switch - AIR COND
Cabin Heat Valve - ON (adjust to desired level)
Fan Switch - HI or LO (as desired)
Outlets - Open overhead windshield outlet and bottom
outlet on instru. panel side duct.**ENGINE SHUTDOWN**Environmental Control Switch - OFF
Cabin Heat Valve - OFF

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SECTION 3: EMERGENCY PROCEDURES

Environmental Control Switch - OFF

Cabin Heat Valve - OFF

For any of the following emergencies:

Fuel control and/or governor failure

Engine Failure

Engine Over Temperature

Insufficient Power

Generator Failure

Note: For 206B model designated TH67 with Two Pilot
IFR: In the event of a main generator failure the air
conditioning system is automatically disconnected.

SECTION 4: MALFUNCTION PROCEDURESCaution Light (Amber) Segments (when heating is
installed)

Duct (Temp High) - ON

1. Environmental Control Switch - FAN
2. Fan Switch - "LO" or "HI"
3. If light continues - Decrease heat or turn
OFF. (This step only when Heater only is
installed.)

(When ECS or Cabin Heating only is Installed.)

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SECTION 5: PERFORMANCE DATA

Environmental Control Switch - AIR COND

NOTE:

Reduce the performance data in the basic flight
manual by the amount given below when the Air
Conditioning is operating and the intersection of
the operating pressure altitude and the OAT is
above the application curve shown below. The
Hover Ceiling Gross Weight is reduced by the
amount given below when applicable in all
conditions.

Rate of Climb	95 ft/min.
Hover Ceiling Gross Weight	68 lbs.

EXAMPLE:

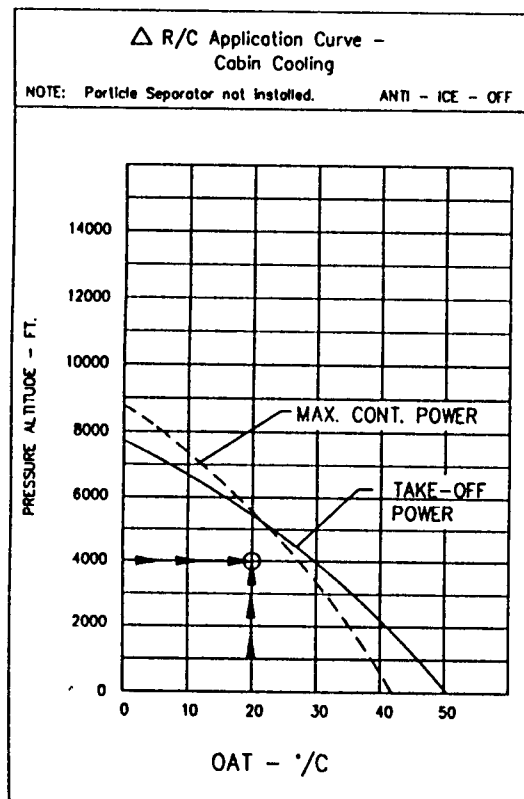
What reduction in Rate of Climb could be expected
under the following conditions:

Pressure Altitude - 4,000 feet
OAT - 20°C

Enter at an OAT of 20°C, move vertically to the 4,000 ft.
Pressure Altitude line. This intersection is below the
curve, therefore, no rate of climb change applies.

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SECTION 5: (Cont'd) PERFORMANCE DATA

Cabin Heating Valve - ON

NOTE:

Reduce the performance data in basic flight manual or optional equipment supplement with the following charts when the bleed air heater is operating. Performance decrements are shown for the standard engine air inlet and for the particle separator induction system.

Complete hover performance is presented herein for the snow deflector, which includes losses due to the particle separator.

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SECTION 5: (Cont'd)

PERFORMANCE DATA

EXAMPLE:

What gross weight loss in hover performance could be expected under the following conditions:

Standard engine inlet

Standard skid gear

IGE hover

Takeoff power

Outside air temp = -15°C

Anti-ice off

Pressure Altitude = 14,000 feet

Using the appropriate IGE chart, enter at OAT (-15°C), move vertically to intersect pressure altitude curve (or outermost curve, whichever comes first), then proceed horizontally to obtain the gross weight loss (170 pounds). Apply this weight loss to the weight obtained from appropriate hover performance chart in basic flight manual or supplement.

There is no loss in hover performance when the outside air temperature is to the left of the pressure altitude curve. It can be seen on the chart covering the above conditions that at -15°C there is no loss in IGE hover performance from sea level to 12,000 feet. Similarly, it can be seen that there is no hover loss below 11,000 feet on a standard day (OAT = -6.8°C).

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HOVER CEILING DECREASE
DUE TO BLEED AIR HEATER OPERATION
STANDARD INLET WITH STANDARD SKID GEAR

IN GROUND EFFECT

TAKEOFF POWER

-40° TO 30°C

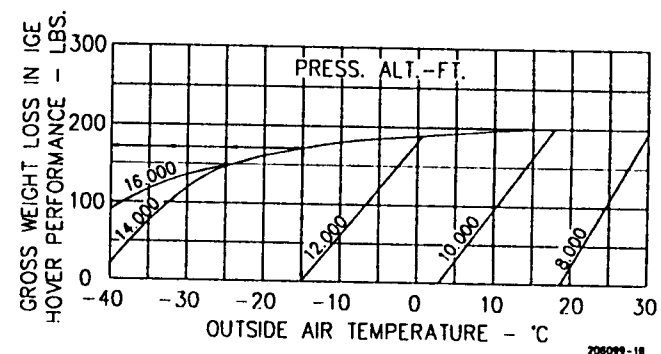
GENERATOR 22.3 AMPS

SKID HEIGHT 2.0 FT (0.6 METER)

ANTI-ICE OFF

ENGINE RPM 100%

WITH ANTI-ICE ON APPLY ADDITIONAL DECREMENT FROM BASIC
MANUAL OR APPROPRIATE SUPPLEMENT



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HOVER CEILING DECREASE
DUE TO BLEED AIR HEATER OPERATION
STANDARD INLET WITH HIGH SKID OR ANY FLOAT GEAR

IN GROUND EFFECT

TAKEOFF POWER

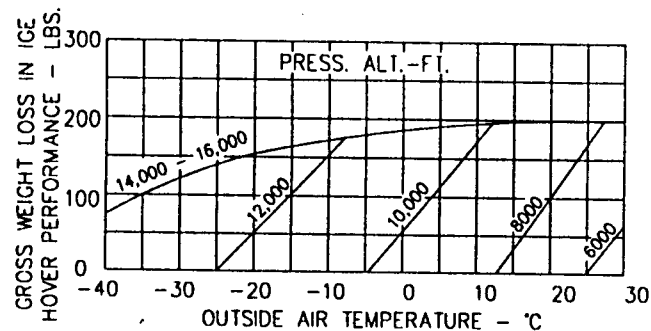
-40° TO 30°C

GENERATOR 22.3 AMPS

SKID HEIGHT 2.0 FT (0.6 METER)

WITH ANTI-ICE ON APPLY ADDITIONAL DECREMENT FROM
APPROPRIATE SUPPLEMENT

ANTI-ICE OFF
ENGINE RPM 100%



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HOVER CEILING DECREASE
DUE TO BLEED AIR HEATER OPERATION
STANDARD INLET WITH ANY SKID OR FLOAT GEAR

OUT OF GROUND EFFECT

TAKEOFF POWER

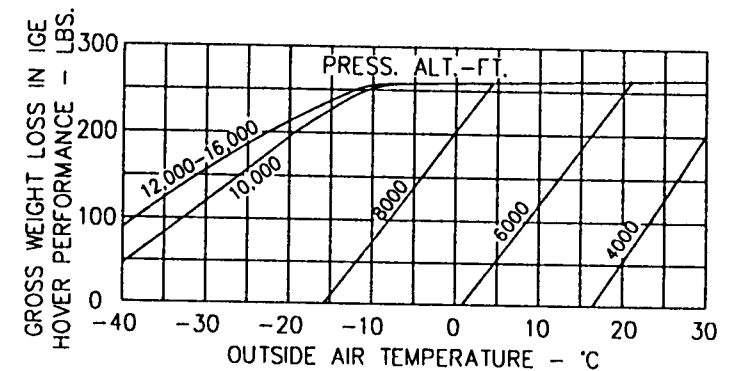
-40° TO 30°C

GENERATOR 22.3 AMPS

SKID HEIGHT 40 FT (12.2 METERS)

WITH ANTI-ICE ON APPLY ADDITIONAL DECREMENT FROM
BASIC MANUAL OR APPROPRIATE SUPPLEMENT

ANTI-ICE OFF
ENGINE RPM 100%



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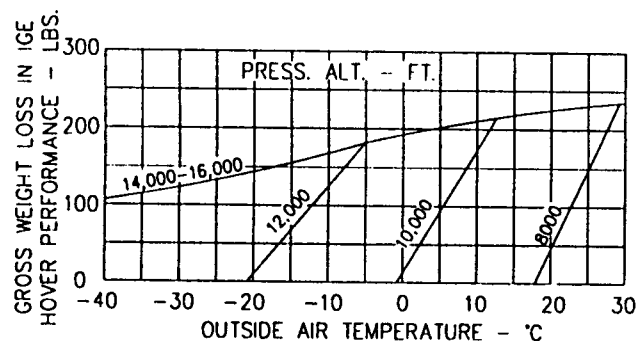
**206B FLIGHT MANUAL SUPPLEMENT
250-C20B/C20J ENGINE**

CABIN COOLING, CABIN HEATING or ECS

HOVER CEILING DECREASE
DUE TO BLEED AIR HEATER OPERATION
PARTICLE SEPARATOR WITH STANDARD SKID GEAR
IN GROUND EFFECT TAKEOFF POWER

-40° TO 30°C

GENERATOR 22.3 AMPS ANTI-ICE OFF
SKID HEIGHT 2.0 FT (0.6 METER) ENGINE RPM 100%
WITH ANTI-ICE ON APPLY ADDITIONAL DECREMENT FROM
PARTICLE SEPARATOR SUPPLEMENT



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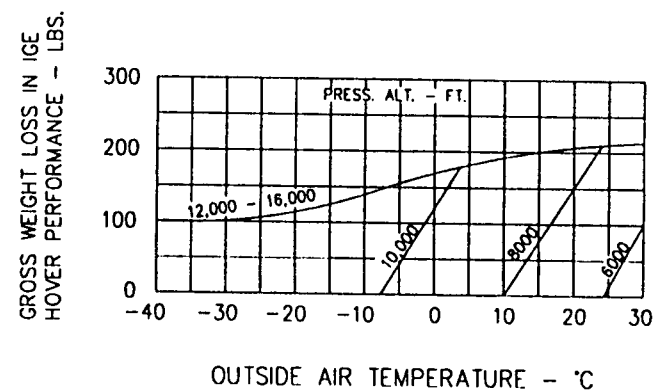
**206B FLIGHT MANUAL SUPPLEMENT
250-C20B/C20J ENGINE**

CABIN COOLING, CABIN HEATING or ECS

HOVER CEILING DECREASE
DUE TO BLEED AIR HEATER OPERATION
PARTICLE SEP. WITH HIGH SKID OR ANY FLOAT GEAR
IN GROUND EFFECT TAKEOFF POWER

-40° TO 30°C

GENERATOR 22.3 AMPS ANTI-ICE OFF
SKID HEIGHT 2.0 FT (0.6 METER) ENGINE RPM 100%
WITH ANTI-ICE ON APPLY ADDITIONAL DECREMENT FROM PARTICLE
SEPARATOR SUPPLEMENT



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CABIN COOLING, CABIN HEATING or ECS

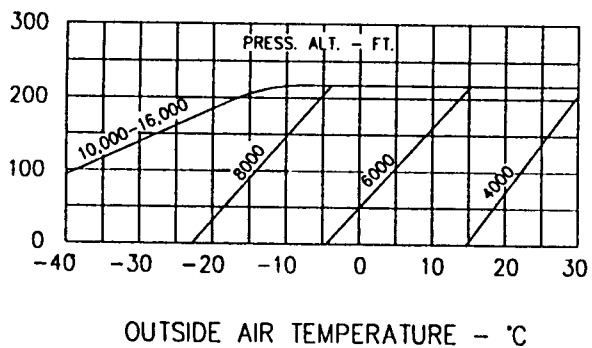
**HOVER CEILING DECREASE
DUE TO BLEED AIR HEATER OPERATION
PARTICLE SEPARATOR WITH ANY SKID FLOAT GEAR**

OUT OF GROUND EFFECT TAKEOFF POWER
-40° TO 30°C

GENERATOR 22.3 AMPS
SKID HEIGHT 40 FT (12.2 METERS)
WITH ANTI-ICE ON APPLY ADDITIONAL DECREMENT FROM PARTICLE
SEPARATOR SUPPLEMENT

ANTI-ICE OFF
ENGINE RPM 100%

GROSS WEIGHT LOSS IN OGE
HOVER PERFORMANCE - LBS.



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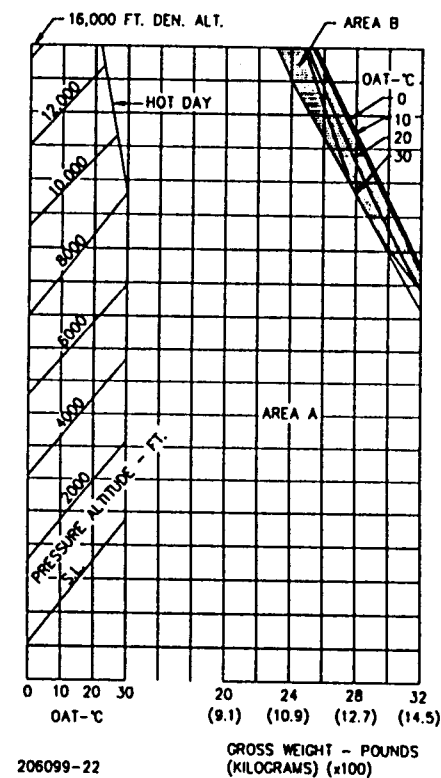
**206B FLIGHT MANUAL SUPPLEMENT
250-C20B/C20J ENGINE**

CABIN COOLING, CABIN HEATING or ECS

**HOVER CEILING
SNOW DEFLECTOR WITH STANDARD SKID GEAR
IN GROUND EFFECT TAKEOFF POWER**
0° TO 30°C

GENERATOR 22.3 AMPS
SKID HEIGHT 2.0 FT (0.6 METER)
WITH ANTI-ICE ON GROSS WEIGHT IS 245 LBS (111.1Kg) LESS

ANTI-ICE OFF
ENGINE RPM 100%

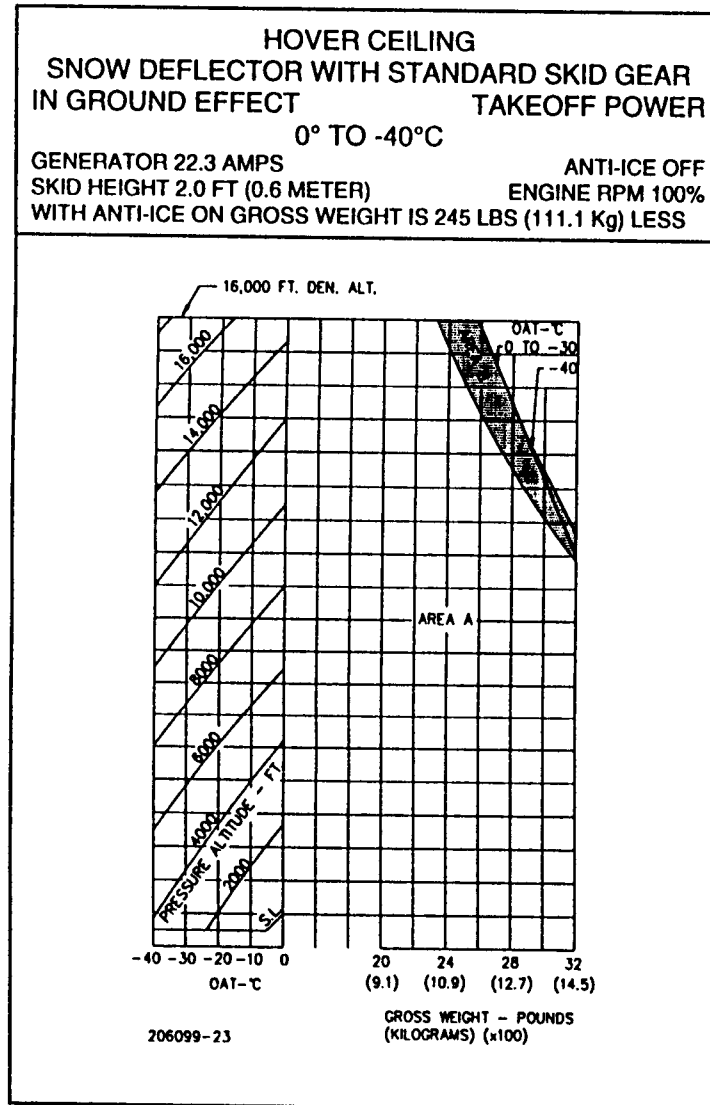


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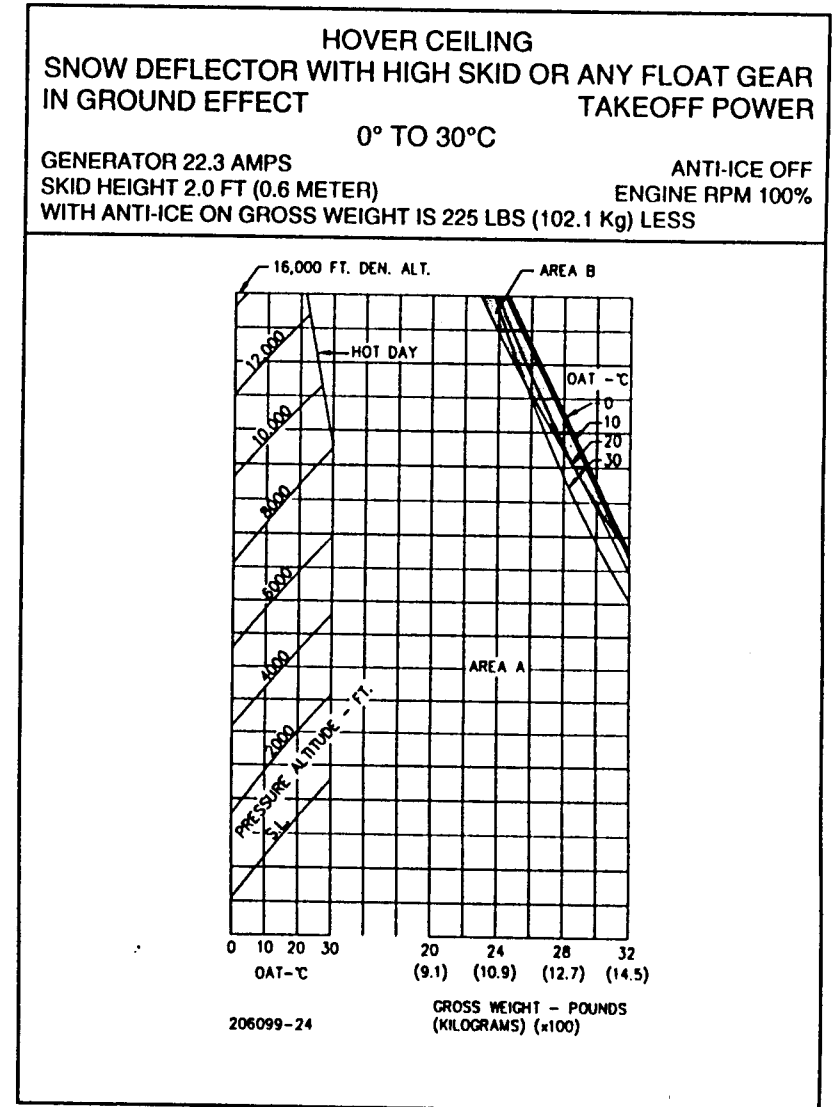
CABIN COOLING, CABIN HEATING or ECS



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**206B FLIGHT MANUAL SUPPLEMENT
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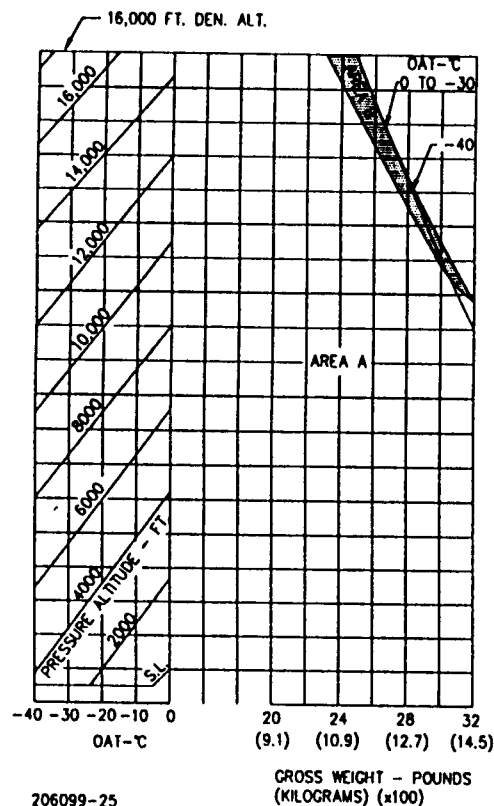


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**206B FLIGHT MANUAL SUPPLEMENT
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CABIN COOLING, CABIN HEATING or ECS

HOVER CEILING
SNOW DEFLECTOR WITH HIGH SKID OR ANY FLOAT GEAR
IN GROUND EFFECT TAKEOFF POWER
0° TO -40°C
GENERATOR 22.3 AMPS
SKID HEIGHT 2.0 FT (0.6 METER)
WITH ANTI-ICE ON GROSS WEIGHT IS 225 LBS (102.1 Kg) LESS
ANTI-ICE OFF
ENGINE RPM 100%

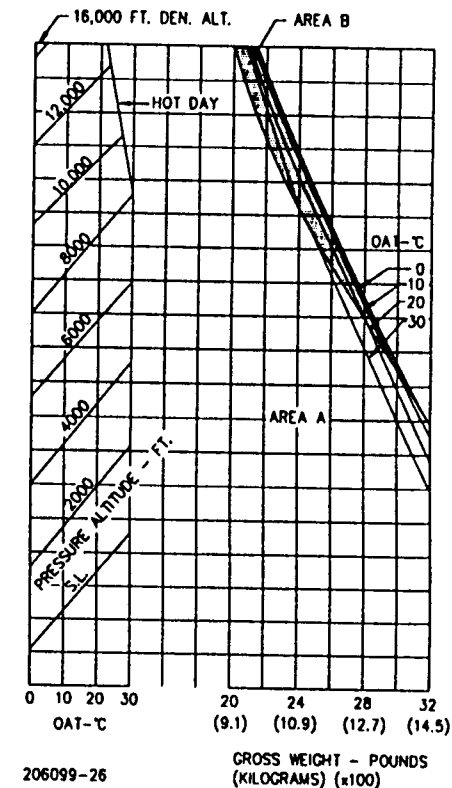


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HOVER CEILING
SNOW DEFLECTOR WITH ANY SKID OR FLOAT GEAR
OUT OF GROUND EFFECT TAKEOFF POWER
0° TO 30°C
GENERATOR 22.3 AMPS
SKID HEIGHT 40 FT (12.2 METERS)
WITH ANTI-ICE ON GROSS WEIGHT IS 260 LBS (117.9 Kg) LESS
ANTI-ICE OFF
ENGINE RPM 100%

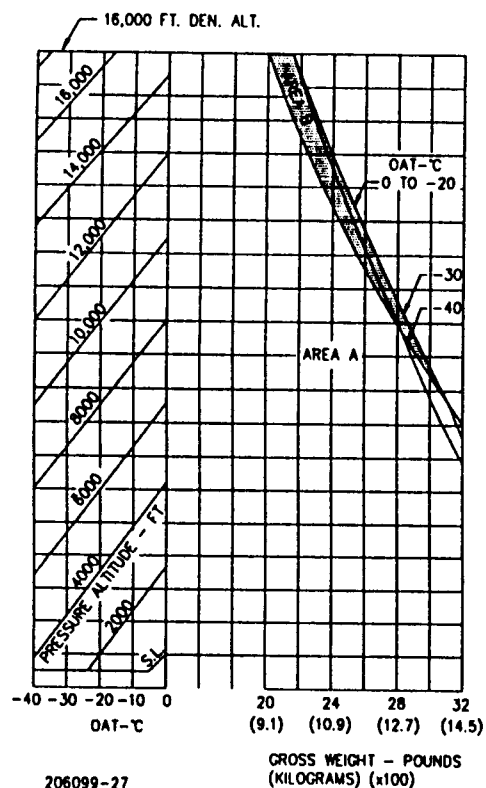


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CABIN COOLING, CABIN HEATING or ECS

HOVER CEILING
SNOW DEFLECTOR WITH ANY SKID OR FLOAT GEAR
OUT OF GROUND EFFECT **TAKEOFF POWER**
0° TO -40°C
GENERATOR 22.3 AMPS **ANTI-ICE OFF**
SKID HEIGHT 40 FT (12.2 METERS) **ENGINE RPM 100%**
WITH ANTI-ICE ON GROSS WEIGHT IS 260 LBS (117.9 Kg) LESS



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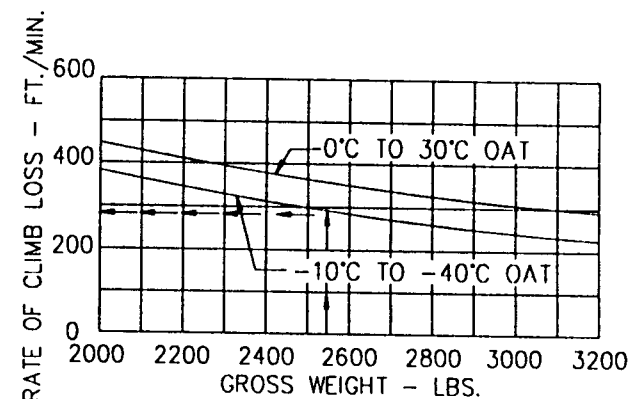
206B FLIGHT MANUAL SUPPLEMENT 250-C20B/C20J ENGINE

CABIN COOLING, CABIN HEATING or ECS

RATE OF CLIMB DECREASE
 DUE TO BLEED AIR HEATER
 ANY INLET WITH ANY SKID OF FLOAT GEAR

TAKEOFF POWER

GENERATOR 22.3 AMPS
V IND 60 MPH (52 KNOTS)
WITH ANTI-ICE ON APPLY ADDITIONAL DECREMENT FROM
BASIC MANUAL OR APPROPRIATE SUPPLEMENT

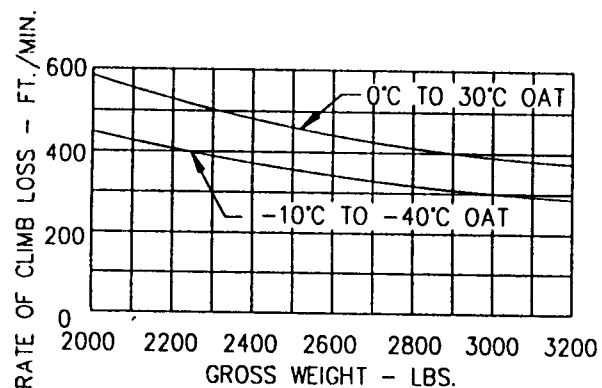


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250-C20B/C20J ENGINE**

CABIN COOLING, CABIN HEATING or ECS

**RATE OF CLIMB DECREASE
DUE TO BLEED AIR HEATER OPERATION
ANY INLET WITH ANY SKID OR FLOAT GEAR****MAXIMUM CONTINUOUS POWER**GENERATOR 22.3 AMPS
V IND 60 MPH (52 KNOTS)ANTI-ICE OFF
ENGINE RPM 100%WITH ANTI-ICE ON APPLY ADDITIONAL DECREMENT FROM
BASIC MANUAL OR APPROPRIATE SUPPLEMENT

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